

Increasing the autonomy of electrical MAV : Design of a new electrochemical generator featuring high specific power and energy

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Abstract

The high energy needed for powering electrical MAVs or UAVs can be obtained by using Aluminium-air Fuel cells, which have a high potential energy density.

Such batteries can reach more than 250 Wh/kg, while lithium-polymer batteries are currently limited to around 160 Wh/kg.

A 100 W prototype is being developed to power small UAVs up to 1000 g.

The technical concepts of the Al/air fuel cell are described and the performances obtained with the first prototypes are discussed.

- This project is co-funded by French Agency DGA / SPNum



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Required Power

Electrical Flight requires high power to mass ratios

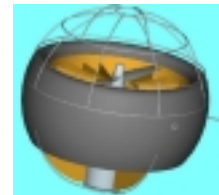
1) For a plane UAV

- 80 W/kg for slow horizontal flight
- 100 W/kg for calm flight
- 150 W/kg for acrobatic flight
- 400 W/kg for vertical climbing
(lower ratios can be required for gliders with low wing loads)



2) For an helicopter UAV

- 320 W/kg for hovering
- 450 W/kg for fast evolution



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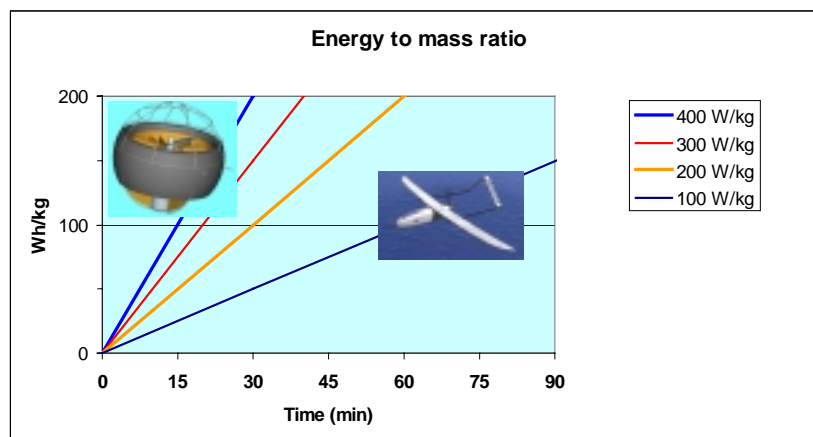
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Required Energy

Endurance requires high Energy to mass ratios



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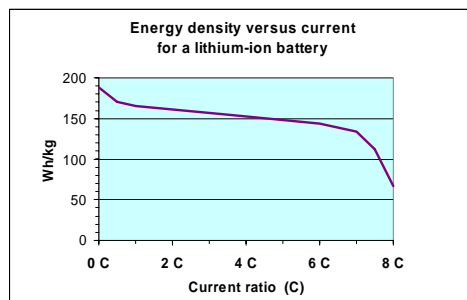


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Battery limits

Best available Lithium-Ion batteries provide :

- 180 Wh/kg at very low current
- 160 Wh/kg at 2 C continuous (discharge in 30 min, i.e. 320 W/kg)
- Decreasing to 130 Wh/kg at very high current (5C to 7C continuous)

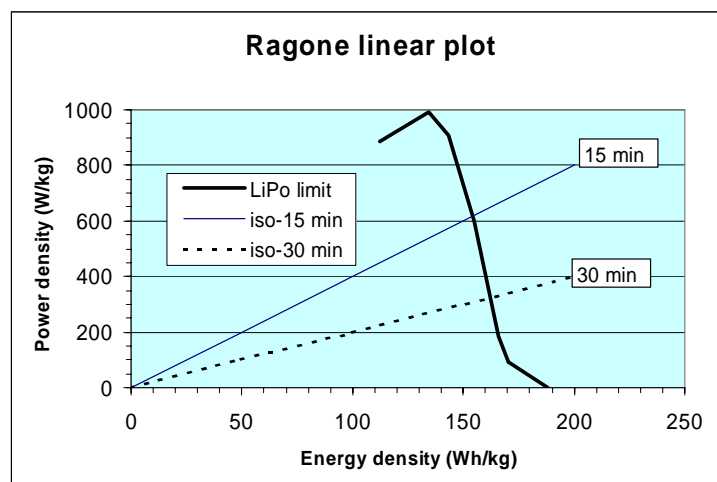


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Battery limits



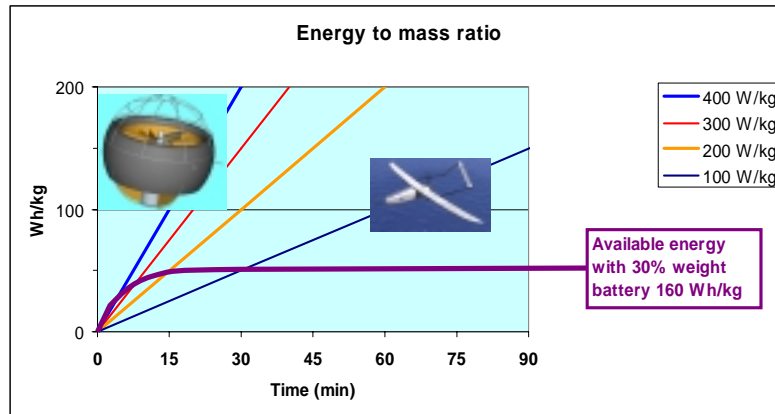
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Battery limits

Best LiPo endurance : 5 to 30 min assuming 30% mass battery/UAV ratio



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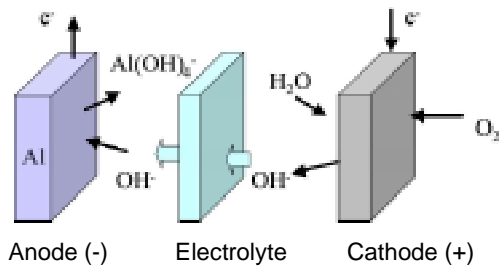
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Aluminium-Air Fuel Cell

Alkaline Metal fuel Cell type

Principle

- Air breathing (cathode side)
- Aluminium consumption (anode)
- Water consumption (electrolyte)



Key Issues

- ◆ Corrosion inhibition (H₂ parasitic production)



- ◆ Cathode optimisation (catalyst)
- ◆ Electrolyte consumption limitation

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Aluminium-Air Fuel Cell

Fuel energy densities

- Theoretical limit for aluminium : 3 Ah/g - 2,7 V
 ➡ 8100 Wh/kg Alu
- Maximal potential 1,6 V, H₂O Consumption 1 kg/kg Alu
 ➡ 2400 Wh/kg fuel [Alu + water]

Fuel Cell Power & Energy densities

- State of the art : Pm = 150 W/kg, Em = 150 Wh/kg
- Possible : Pm = 500 W/kg, Em = 400 Wh/kg

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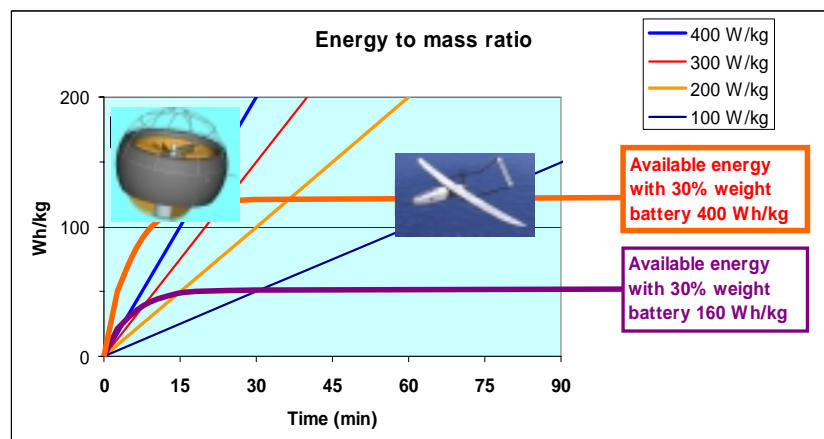
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Aluminium-Air Fuel Cell

DROPAC project



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Aluminium-Air Fuel Cell

Former Research

- **Zaromb cell** & US Philco Company (1960) : 500 Wh/kg - 1 A/cm²
- **DESPEC** (1985) with saline electrolyte (Al/O₂ for submarines) (commercialized by **ALUPOWER** --> **Fuel Cell Tech**)
- **ALCAN** --> **Yardner** (1989) : 13% efficiency
- **Electro-Chem-Technic** UK 1997 : small cells for education (No longer available)



Electro-Chem-Technic, Oxford, (UK)

Manufacturers

- **Aluminum Power, Inc. (Canada)** --> **Trimol 2000** 800 Wh/kg & 70 mA/cm² claimed (No news available)
- **Europositron 2003**, project of a rechargeable system : 60 kg, 80 kWh (1330 Wh/kg) objectives (no products)



Trimol / Aluminum - Power Inc. Toronto (prototype)

--> No commercially available products

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Aluminium-Air Fuel Cell

DROPAC project (CEA + Bertin)

- **Power generator for MAV or UAV electrical propulsion**
 - ❖ Low mass, i.e. high Power density (> 500 W/kg)
 - ❖ Surpass chemical batteries (Li-Ion) (>> 160 Wh/kg)
- **Performance Objectives**
 - ❖ 75 W continuous, 100 W maximum power
 - ❖ Mass < 200 g
 - ❖ Operation time > 30 min

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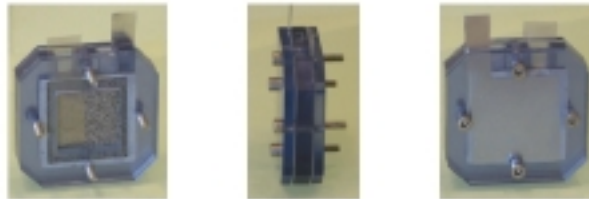
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Aluminium-Air Fuel Cell

DROPAC prototype n°1 : 9 cm² active surface

- for basic experiments



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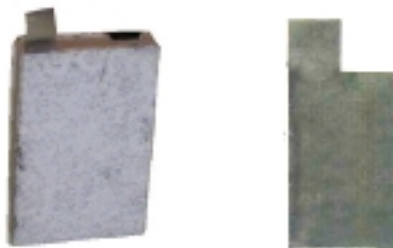
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Aluminium-Air Fuel Cell

**DROPAC prototype n°2 & 3 : 47 cm² active surface
10 W - 1 V**



Current achievements :

- Up to 30 min at 7,5 W, 10 W maximum
- Mass 20 g
- Power and Energy ratios : 500 W/kg max, \approx 190 Wh/kg

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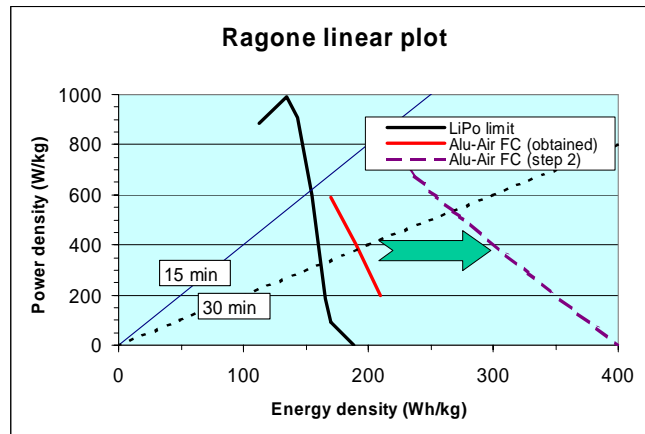
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Aluminium-Air Fuel Cell

DROPAC prototype achievements and objectives



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Aluminium-Air fuel cell

Current development

- Current tests on a complete 10-cells generator (100 W peak), for airborne tests (on a 1 kg model plane)



- Laboratory tests for upgraded performances

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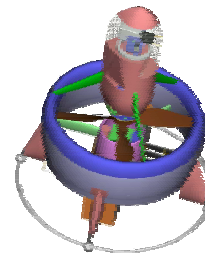
Aluminium-Air fuel cell

Future developments

- Reusability (anode + electrolyte change)
- Improve usability (ready-to-use system)

Foreseen applications

- UAV Electrical Propulsion (helicopter or plane types)
- Other systems requesting short duration high power



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